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\ miter using this principle can be conat any frequency, depending only availability of an adequate de magacld, whose amplitude is somewhat than the ferrimagnetic resonance Theoretical considerations show that siling factor, which is proportional to the ply volume divided by the cavity vol-, west be large in order to achieve limitover a significant range of power. The Jability of large single crystals of ytin iron garnet now makes it possible to thin these large filling factors along with ow linewidths.

Fig. 1 is a schematic of a 4 kMc model this limiter which used a polished sphere ingle crystal YIG, 0.260 inches in dieter. The desired limiting value was 14 n. This level can be adjusted to as low 3 dbm with the dc magnetic field. A Jaced-height (0.400 in) transmission cavwith a loaded Q of 200 was operated in TE₁₀₁ mode with the sample placed in e maximum RF magnetic field close to the ic wall of the cavity. At low power levels e sample has very little effect on cavity msmission properties. Above a well-deed threshold level sample losses increase pidly and limit the transmission power vel.

Temperature changes can have large fects on the characteristics of this device. he limiting power level changes with $4\pi M$, M_k , H_{de} and H_a , the anisotropy field. In Idition, there are low-level insertion loss ariations due to thermal cavity detuning fects. These include cavity expansion and RF susceptibility variations dependent on $-M_{i}$ $\Delta H_{k_{i}}$ H_{de} and H_{a} . A major source of thermal changes in limiter performance can e avoided by the use of a spherical sample, or which the ferromagnetic resonance frequency is independent of the saturation rignetization and its associated temperature variation. It can also be shown that rientation of the anisotropy field can be sed to minimize further the temperature ensitivity. The samples used here were of -purity such that ΔH_k had only a slightly egative temperature coefficient at room emperature. For these samples orientation the magnetically hard [100] crystallo-caphic axis along the biasing field gave plinum temperature characteristics.

The limiter performance over the design temperature range from 55° to 120°F is shown in Fig. 2. The sharp break in charatteristic at the threshold is important for r'heient power leveling. It is dependent on he use of single-crystal narrow-linewidth material and the surface polish. In this case Mk was approximately 0.15 oe at 4 kMc. pike leakage was observed but not inestigated; however, this limiter should have haracteristics similar to the coincidence uniter2 in this respect.

Because of weight limitations in the stellite application, the cavity was fabriited from silver-plated magnesium resultin a total weight, including magnet, of " ourses. While the limiter was designed at pa ticular frequency with particular rearements, it should prove useful in other

¹ J. F. Dillon, Jr., "Ferrimagnetic resonance in attium iron garnet," Phys. Rev., vol. 105, pp. 759-766; January, 1957.

applications at different frequencies requiring precise control of power level.

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Power-Aperture and the Laser*

The author has heard frequent reference to the power-aperture product criterion for radar surveillance, but has not yet seen in print a reasonably general proof of the relation, as simple as it is. The situation is particularly important at present with the development of the laser, its very narrow beamwidth and the consequent possibility of a very long range optical radar with a small aperture. It is necessary, therefore, to consider the limitations of this device, particularly for surveillance purposes. To anticipate the result: the surveillance of a given number of square degrees of coverage per second out to some range with a given input noise, requires at least the same average power-aperture product at optical as at microwave frequencies.

LARGE SPOT SIZE

We consider first the usual radar case in which the beamwidth is much larger than the target (large spot size). $E_T = P_{TT}$ is the transmitted energy in an element of solid angular resolution ΔΩ during a total coherent time of integration τ ; the total coherent time r may be broken up and distributed over a longer time interval provided the returns from an individual target can be processed coherently. P_T is the average transmitted power during time τ and corresponds to the peak pulse power if the power is emitted in uniform rectangular pulses. For a CW fence radar, \u03c4 would ideally correspond to the time of passage of the target through the fence. The transmitting antenna gain

The basic radar equation, in terms of the returned energy, E_R , in time τ , is

$$E_R = \frac{E_T G_T \sigma A_R}{(4\pi R^2)^2} = \frac{E_T \sigma A_R 4\pi}{(4\pi)^2 R^4 \Delta \Omega}$$

where A_R = receiving antenna aperture area. At the output of a matched filter, the signal-to-noise energy or power ratio is $2E_R/N$, where N is the input noise power per cps of bandwidth (noise spectral density).

Therefore, if we denote by C the signal-tonoise power margin required for some preassigned probability of detection,

 $C = \frac{2E_R}{N} = \frac{2E_T \sigma A_R}{4\pi N R^4 \Delta \Omega}.$

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If n is the total number of elements e_n angular resolution of the transmitting antenna swept out during a surveillance scan interval t and if Eq is the total energy radiated during t, then $nE_T = E_G$, and

$$C = \frac{2E_G\sigma A_R}{4\pi NR^3n\Delta\Omega} = \frac{2E_G\sigma A_R}{4\pi NR^4\Omega}; \quad n\Delta\Omega = \Omega$$

 $E_G = P_A t$, where P_A is the average transmitted power in a scan interval, so

$$C = \frac{2P_A l \sigma A_R}{4\pi N R^4 \Omega} \text{ or } 2\pi N C \frac{R^4}{\sigma} \left(\frac{\Omega}{\ell}\right) = P_A A_R$$

This is the surveillance equation for large spot size, and shows that the solid angular coverage swept out per second is proportional to the product of the average transmitted power and the receiving aperture.

There are, of course, other limitations not expressed in the previous equation, such as mechanical limitations on speed of solid angular coverage, and assurance that the receiving aperture is in the correct angular position to receive the return echo at the time of arrival.

In an actual radar there will be losses, which can be lumped in with C. An important corollary is obvious from the above theorem, by virtue of the fact that (ΣP_i) . $(\Sigma A_i) > \Sigma P_i A_i$, for i > 1. For a given rate of coverage out to a given range for the case of separate radars, ot synchronized or coherent with one another, the smaller the number of radars, the smaller the total poweraperture requirements; in fact, if line-of-sight requirements permit, one radar is the most efficient. For example, if we took an extreme case of 100 radars, then the system would require ten times as much total aperture area and ten times as much total average power as would the single radar installation.

With a laser radar, in the case of large spot size, the same fundamental coverage considerations apply as for a microwave radar. Therefore, unless the assumptions of the foregoing analysis can be controverted for the laser, since we will need as high power and as large aperture for the optical as for microwave radar, there is little advantage to the laser for long range surveillance, and we will probably continue into the indefinite future with radars in the general microwave region for this purpose. Of course, in tracking with large spot size, it is true that a laser radar with a small aperture can track to the same range as a microwave radar with equal power and noise and much larger aperture.

SMALL SPOT SIZE

It has been pointed out that the laser is capable of such narrow beamwidth that the spot size can be smaller than the target, so that all the power in the transmitted beam is intercepted by the target, resulting in an inverse-square optical radar range relation as compared to the inverse-fourth for the microwave radar. This inverse-square relation may be true for relatively short ranges, but at a long range such as 1000 nautical miles, for an optical-wavelength laser radio with a 1-cm aperture, the spot size due to the diffraction-limited beamwidth is about 300 feet across. A large percentage of space objects will be considerably less than this. Thus, for space applications over appreciable ranges, for the most part, with the small apertures presently discussed for laser radars, the spot sizes will be large, and the conventional inverse-fourth power radar quation holds. Laser radar apertures will have to be raised to linear dimensions of the order of meters rather than centimeters to derive what advantages there are from small spot size. However, once we get to the larger dimensions cited, one of the main advantages of the optical radar, small size, is lost. However, for shorter ranges, such as 10 nautical miles, a tremendous power or aperture advantage is theoretically available over a microwave radar subject to the inverse-fourth power restriction.

It is important, though, to realize that in the whole preceding discussion we have given minimum attention to target cross section as a function of wavelength and spot size, and the effect of motion and consequent fluctuations. The whole subject is too involved to be treated sketchily as it must in a contribution such as the present one. However, one simple example will be given, to bring out the kind of problem that exists with small spot size.

Suppose that one had a 100-foot perfeetly-reflecting smooth sphere. The only region of the sphere which would reflect back to the source would be a small area surrounding the radius vector from the source to the center of the sphere. Any other region of the sphere would be at such an angle to the radius vector that it would reflect signal away from the source. With large spot size, because of beamwidth much greater than the target, as long as the sphere is within the beam there will always be a radius vector within the beam pc pendicular to some part of the sphere, so that some portion of the radiated energy is bound to be reflected back. However, with small spot size, unless the spot is in the region of the sphere perpendicular to the radius vector, there will be essentially no reflection back to the source; the energy will be specularly reflected in some other direction. The return signal strength is tremendous when the beam is exactly centered on the sphere, but goes essentially to zero when off-center.

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Parametric Amplification and Oscillation at Optical Frequencies*

Recent experimental work1 using intense optical fields produced by a maser has indicated that it is possible to obtain variable parameter interaction in a solid. It follows that the processes of amplification and oscillation utilized in microwave devices may be extended to the optical frequency range. Specifically, we propose that coherent optical

* Received by the IRE, February 5, 1962.

P. A. Franklin, et al., "Generation of optical harmonics," Phys. Rev. Lett., vol. 7, p. 118; August, 1961.

energy may be generated at subfrequencies if a nonlinear dielectric material is driven by an optical maser "pump," such as a ruby. Here we derive the conditions for oscillation in a simple resonant s, stem based on the observed experimental data for second harmonic generation.1-8

Consider a resonant structure composed of two parallel, highly reflecting surfaces bounding a medium of nonlinear dielectric material, such as quartz. We shall excite the medium with a traveling-plane wave of frequency f_p and by parametric excitation produce standing waves in the medium at frequencies f_s and f_i subject to the condition that $f_p = f_s + f_i$. Here it is assumed that the thickness of the medium l is such that there exist resonant modes at f_s and f_i . It is also assumed that the reflectivity of the walls is small at the pump frequency so that the pump wave may propagate through the structure without appreciable reflection. Under these conditions, it may be shown4 that the rate of change of amplitude of the signal wave E_s due to interaction of the "idler" wave E_i with the pump is given by

$$\begin{split} \frac{\partial E_s}{\partial t} &= \frac{1}{4\epsilon l} \int_0^l e^{-jk_s z} \left(\frac{\partial P_s}{\partial t} \right) dz \\ &= \frac{j \omega_s \gamma_{si} E_i * E_p}{4\epsilon l} \int_0^l e^{(j(k_p - k_s - k_i)z} dz \quad (1) \end{split}$$

where k_s , k_i , and k_p are the respective wave vectors $(2\pi/\lambda)$ for the signal, idler, and pump; and it has been assumed that the polarization of the nonlinear medium at frequency f_s is

$$|P_s| = \gamma_{si} |E_i| |E_p| \qquad (2)$$

where γ_{si} is a function of the three frequencies. Taking into account the Q of the cavity at the frequency f_s we obtain the equation.

$$\frac{\partial E_s}{\partial t} = \alpha_{si} E_i^* - \frac{\omega_s}{2Q_s} E_3, \tag{3}$$

and a similar equation for the idler wave,

$$\frac{\partial E_i}{\partial t} = \alpha_{is} E_s^* - \frac{\omega_i}{2O_i} E_i \tag{4}$$

where

$$\alpha_{si} = \frac{j\omega_s \gamma_{si} E_p}{4\epsilon l} I_{si}(l) \tag{5}$$

and $I_{si}(l)$ is the "coherence" integral of (1). Now for oscillation, the rate of growth of the signal and idler waves should be zero or greater, and setting (3) and (4) equal to zero vields

$$\alpha_{is}\alpha_{si}^* - \frac{\omega_i\omega_s}{4O_sO_i} = 0. \tag{6}$$

Using the Q of a planar cavity with power reflectivity R given by

² J. A. Giordmaine, "Mixing of light beams in crystals," *Phys. Rev. Lett.*, vol. 8, p. 19; January, 1962, ² P. D. Maker, *et al.*, "Effects of dispersion and focusing on the production of optical harmonics," *Phys. Rev. Lett.*, vol. 8, p. 21; January, 1962, ¹ R. H. Kineston and A. L. McWhorter, "Perturbation Theory for Parametric Amplification," presented at the PGMTT National Symposium, San Diego, Calif., May 9-11, 1960.

$$Q_s = \frac{k_s l}{1 - R};$$

and setting $\omega_a \cong \omega_i \cong \omega_p/2$, we obtain

$$\frac{\omega_{s}^{2}\gamma_{si}^{2}E_{p}^{2}I_{si}^{2}(l)}{4\epsilon^{2}l^{2}}>\frac{\omega_{s}^{2}(1-R)^{2}}{k_{s}^{2}l^{2}}$$

$$\gamma_{si}^{2}E_{p}^{2}I_{si}^{2}(l) > \frac{4\epsilon^{2}\omega_{s}^{2}(1-R)^{2}}{k_{s}^{2}}$$

as the condition for oscillation at the tre quencies f_i and f_i .

In a similar manner we may calculathe second harmonic electric field for traveling wave of frequency f_p obtaining

$$E_{2p} = j\omega_p \sqrt{\frac{\mu}{\epsilon}} \gamma_{2p} E_p^2 \int_0^1 e^{j(2k_p - k_{2\mu})t} dt$$

where the polarization at frequency 2/, given by

$$|P_{2p}| = \gamma_{2p} |E_p|^2, \qquad (1)$$

and the generation takes place over a parlength l. We now define the efficiency of a ond harmonic power generation as

$$\eta = |E_{2p}|^2 / |E_p|^2 = \omega_p^2 \frac{\mu}{\epsilon} \gamma_{2p}^2 E_p^2 I_{2p}^2 (l)$$
 (1.

with $I_{2p}(l)$, the "coherence" integral is (10). For a practical experiment, with the proper choice of materials, the values of and the coherence integral I(l) should !approximately the same for second h. monic generations as for parametric mixit, Thus, the inequality of (9) reduces to

$$\eta > (1-R)^2$$

for the same length, l, and pump and tude E_p . Recent experiments¹⁻³ indithat η can be of the order of 10^{-6} indicate that the reflectivity of the cavity was should be 99.9 per cent or higher for oscilla tion. This reflectivity should be obtain: using multiple dielectric layer films; his pump fields using advanced techniq should relax the above requirements.

We have here considered a special of subfrequency generation using a sircavity geometry and a traveling pump w There are many other possible config tions for such cavities utilizing a starwave pump, for example, or taking adv tage of the tensor properties of the cry to obtain longer interaction lengths, sudescribed by Giordmaine² and Maker.³ felt that the possibility of coherent gen tion of lower frequencies as shown by calculation offers great promise as an ternative source of long wavelength to at frequencies where direct maser actinot feasible. In addition, upon the atability of continuous high-power r sources, amplifiers may also be better the above techniques. Experiments are underway to verify the above predicts

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